



**Turbomachinery Laboratory, Texas A&M University
Mechanical Engineering Department**

MICRO TURBOMACHINERY Applications

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MICROTURBOMACHINERY

Justification

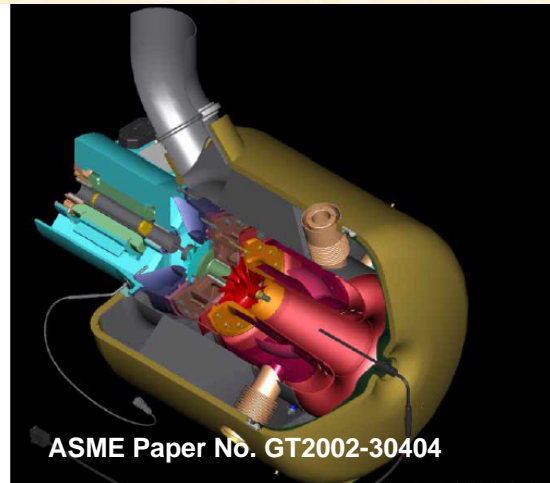
DOE, DARPA, NASA interests range from applications as portable fuel cells (< 60 kW) in microengines to midsize gas turbines (< 400 kW) for distributed power and hybrid vehicles.

Meso-scale or MEMS turbomachinery (< 100 W) for Next Generation Land Warriors, Micro vehicles & robots, Portable electronic devices and systems, Smart munitions



MICROTURBOMACHINERY as per IGTI

Drivers:
deregulation in
distributed
power,
environmental
needs,
increased
reliability &
efficiency



Distributed power
(Hybrid Gas
turbine & Fuel Cell),
Hybrid vehicles

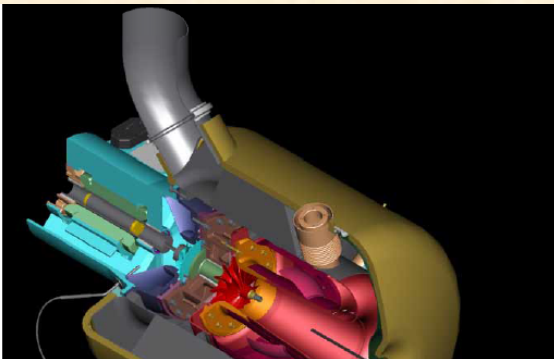
Automotive
turbochargers,
turbo expanders,
compressors,

Max. Power ~
250 kWatt

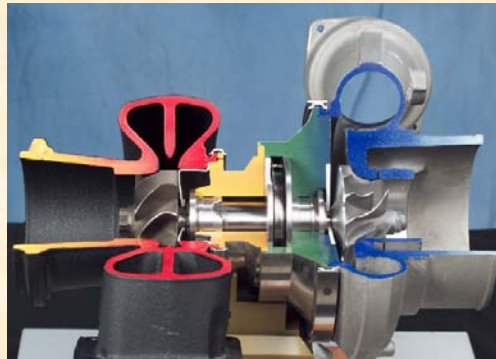


POWER RANGE

< 400 kW



**Distribute power
(Gas turbine & Fuel
Cell Hybrid)**



**Auto engine and
part / Industrial
compressor**



Honeywell, Hydrogen and Fuel Cells
Merit Review



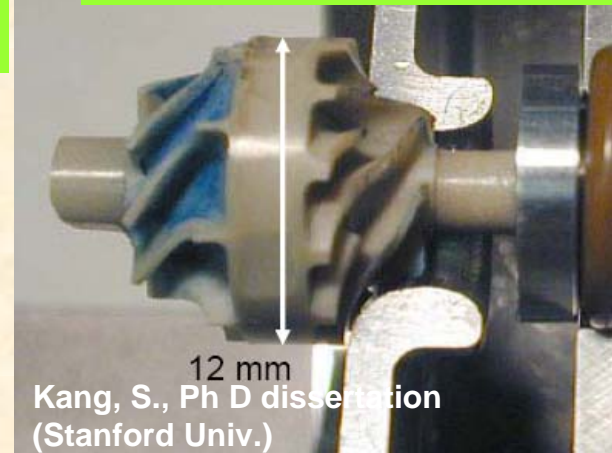
[http://www.miti.cc/newsletters/150
hpcompresso.r.pdf](http://www.miti.cc/newsletters/150_hpcompresso.r.pdf)

< 100 W



[http://smarteconomy.typepad.com/s
mart_economy/2006/09/microgas_tu
rbin](http://smarteconomy.typepad.com/smart_economy/2006/09/microgas_tu_rbin)

**Portable Electronic
Devices**



12 mm
Kang, S., Ph D dissertation
(Stanford Univ.)



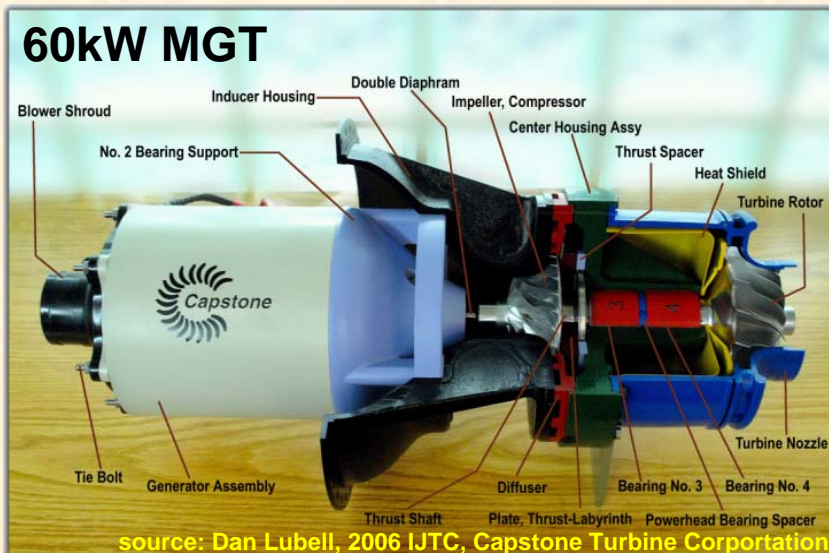
MICRO GAS TURBINES

Cogeneration systems with high efficiency

Microturbine Power Conversion Technology Review, ORNL/TM-2003/74.

- Multiple fuels (best if free)
- 99.99X% Reliability
- Low emissions
- Reduced maintenance
- Lower lifecycle cost

MANUFACTURER	OUTPUT POWER (kW)
Bowman	25, 80
Capstone	30, 60, 200
Elliott Energy Systems	35, 60, 80, 150
General Electric	175
Ingersoll Rand	70, 250
Turbec, ABB & Volvo	100

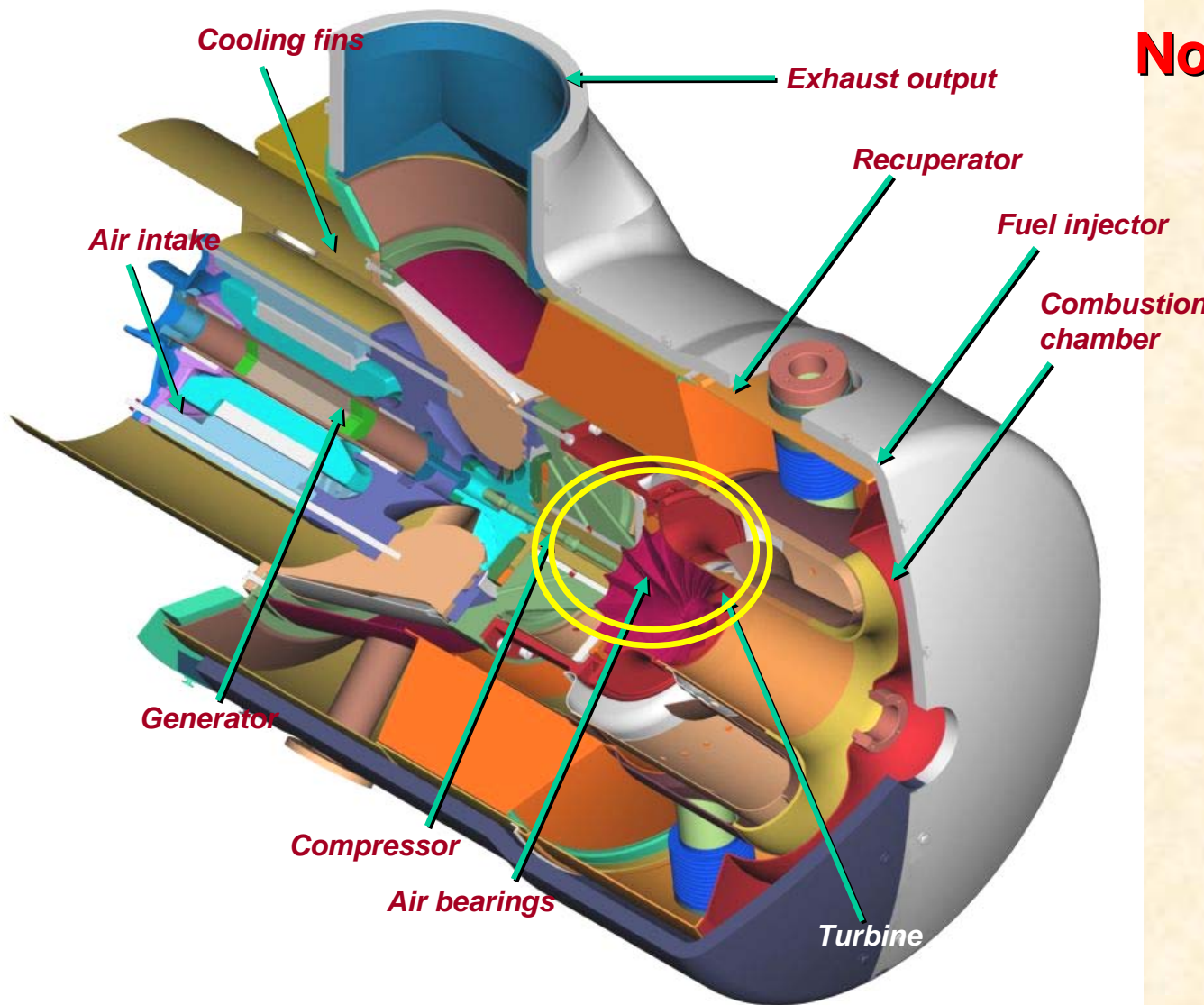


Hybrid System : MGT with Fuel Cell can reach efficiency > 60%

Ideal to replace reciprocating engines. Low footprint desirable



Capstone MicroTurbine™



No gearbox or other mechanicals

Low scheduled maintenance

Only one moving part

No coolants or lubricants

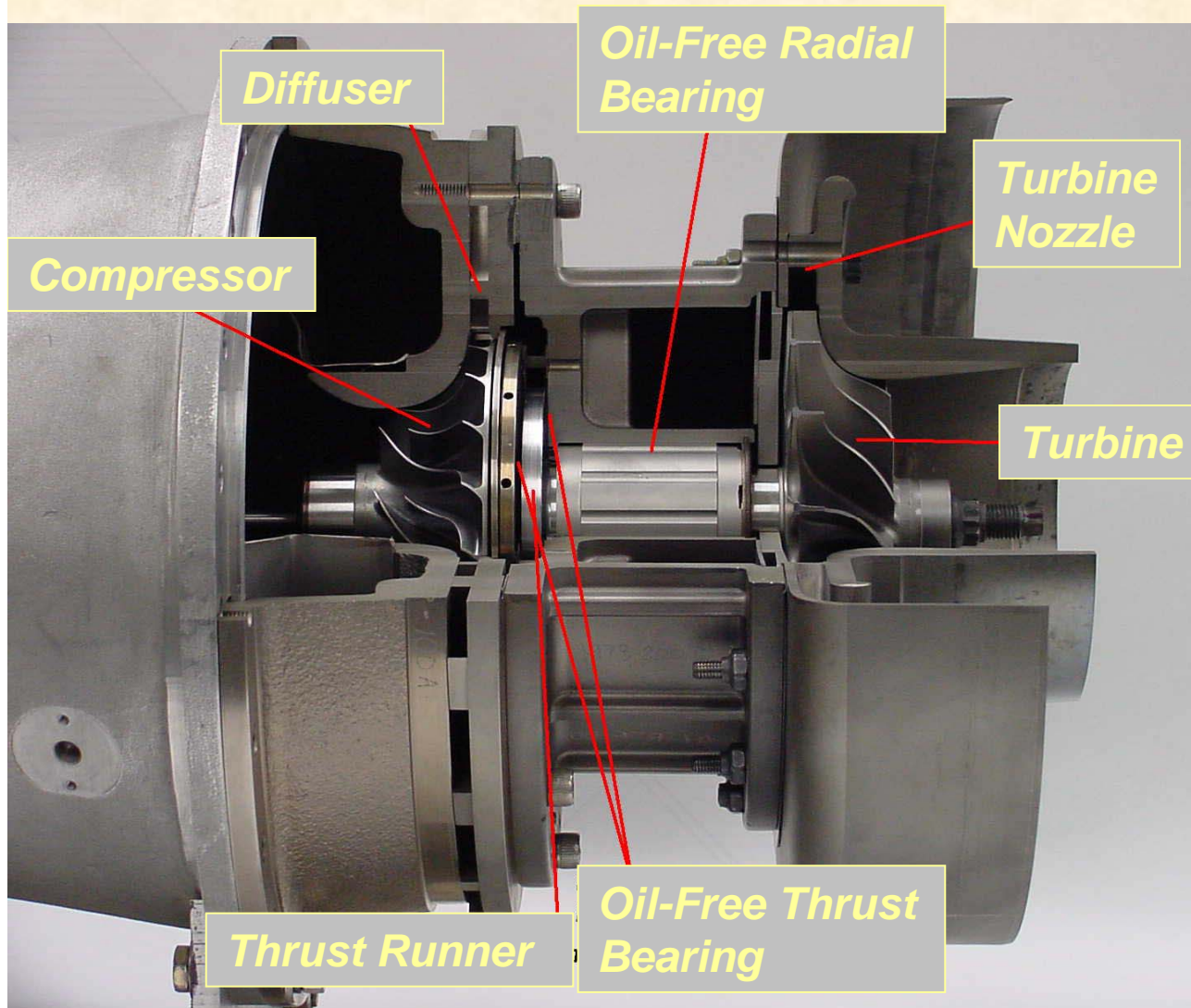
Contaminant-free exhaust

Compact and lightweight

Super-low CO & NO_x



Capstone's C30 Engine



Oil-Free Foil Bearings:

>500°C

Proprietary bearing design and coating

Thin Dense Chrome journals

1.4 MDN (idle)

3.1 MDN (full speed)

~1.5 L/D

1.6 psi static load

Demonstrated Life:

>40k hours; >6k

cycles and over 11

Mhrs field life



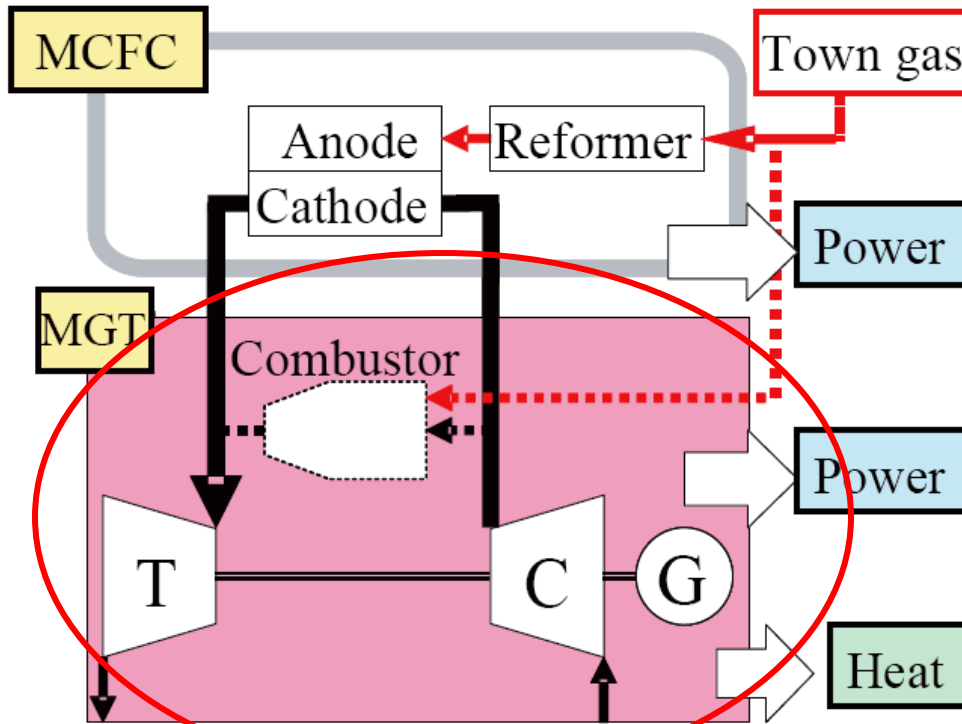
Expectation & Requirement

- **Low cost** – driven by materials
- **Low maintenance** – driven by design
- **Long life** – defined by the bearings and materials
- **Efficient** – driven by design
- **Fully integrated solutions** – system design

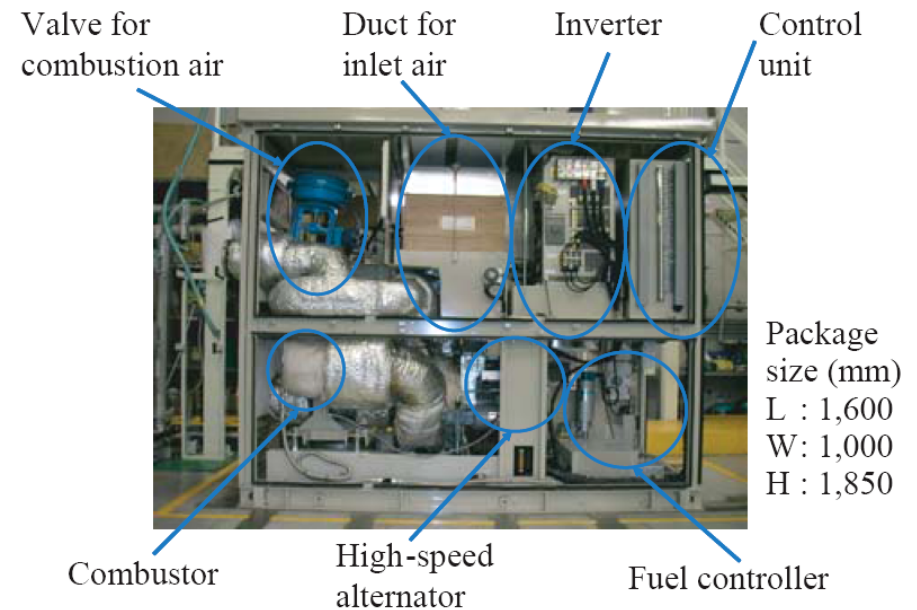


HYBRID GENERATION SYSTEM

MCRC (molten carbonate fuel cell) MT generator



Pressured, and Powered by reformed fuel and air supplied by compressor of MGT



R&D Review of Toyota CRDL, 41

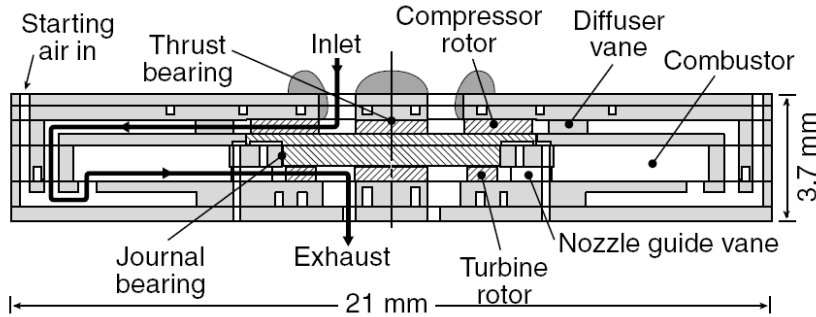
Single-shaft gas turbine (max. 80 krpm)

R&D Review of Toyota CRDL, 41



ULTRA MICROTURBOMACHINERY

MEMS MTM

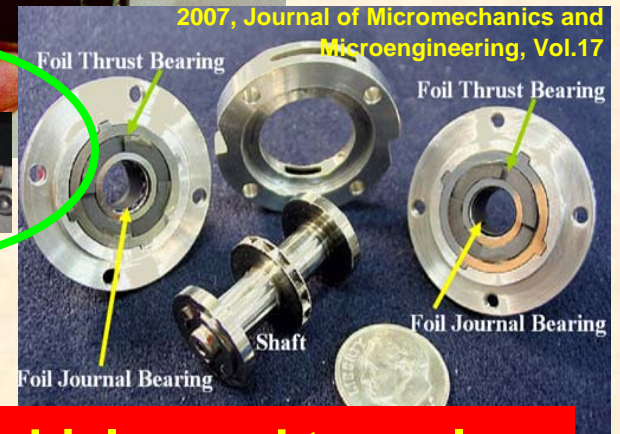


GT-2003-38866

- Silicon wafer
- 1.2 Million rpm
- Thrust 0.1 N
- Spiral groove and hydrostatic gas bearings

Meso-scale MTM

- Palm-size power source
- Brayton cycle
- Gas foil bearings

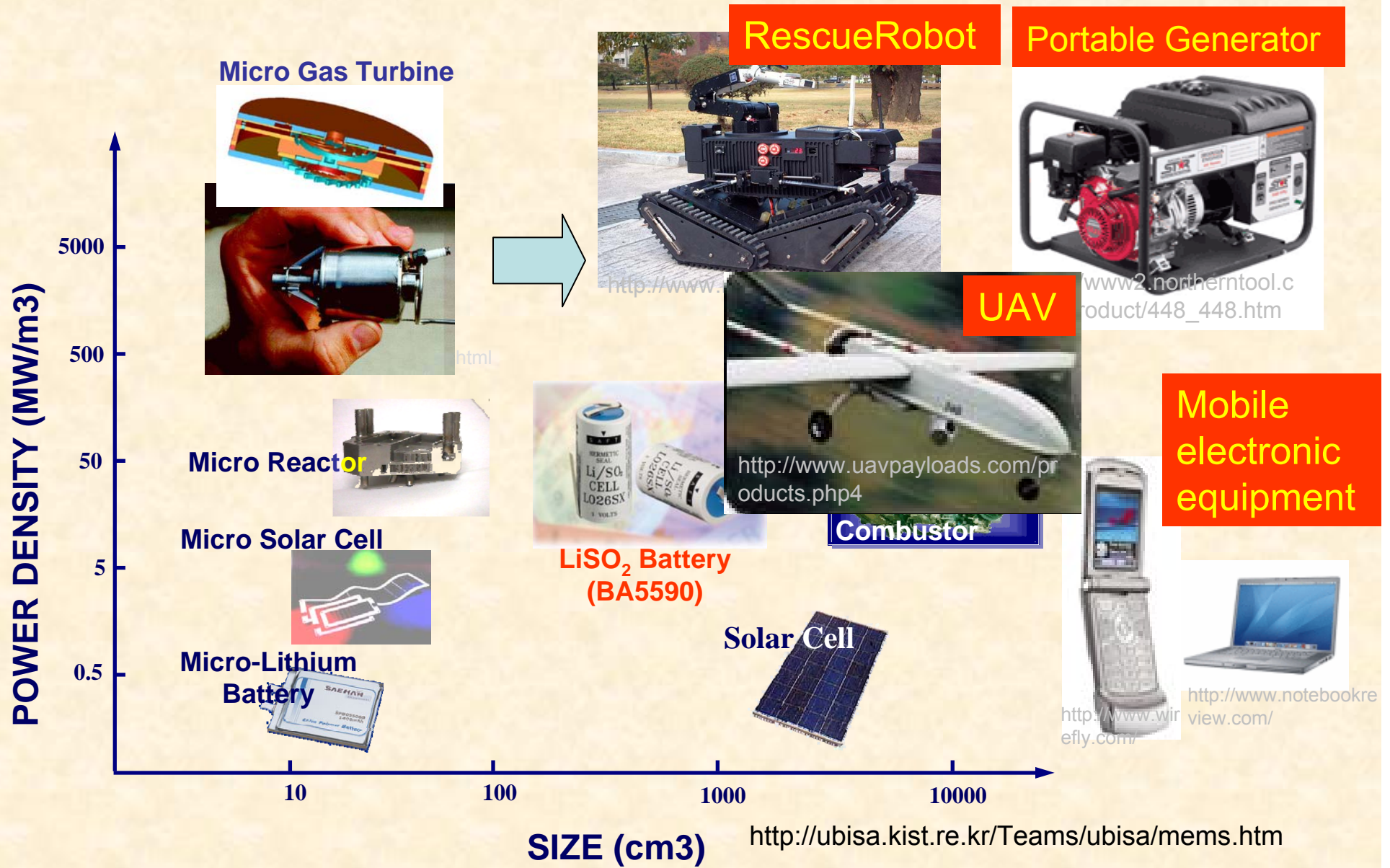


100 Watt & less

Small unmanned vehicles and to replace batteries in portable electronic devices



Application of Meso/MEMS MTM





MEMS MTM at MIT

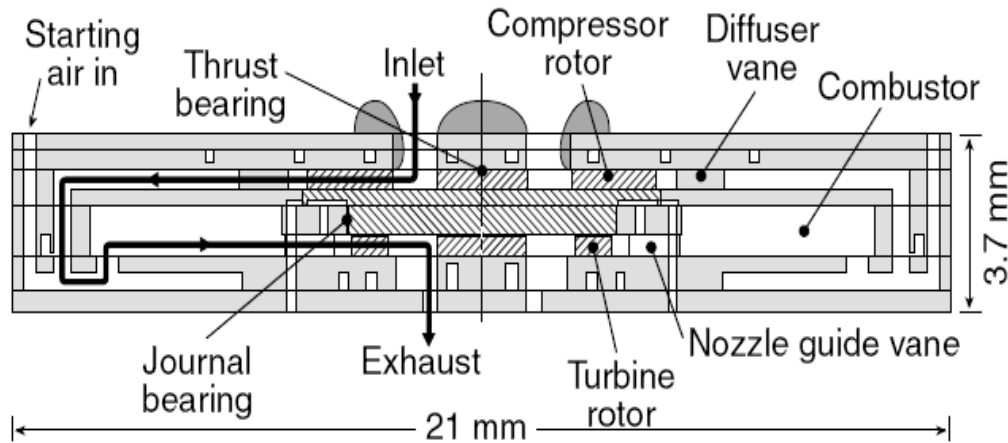


Figure 3: H₂ demo engine with conduction-cooled turbine constructed from six silicon wafers.



Figure 10: A 500 m/s tip speed, 8 mm dia centrifugal engine compressor.

Thrust: 11g (17 watts)

**Turbine inlet temp
: 1600 K**

Fuel burn: 16 gram/hr

Rotor Speed: 1.2 M rpm

Weight: 2 grams

**Exhaust gas temp
: 1243 K**

Source: GT2003-38866



Mesoscale MTM at Stanford

~1997: DARPA – M-Dot project

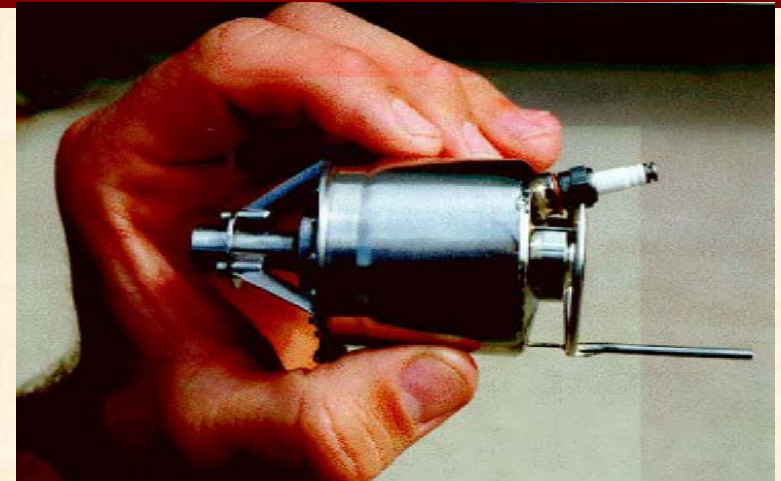
Palm size gas turbine engine (thrust type)

ϕ 25 mm turbine, 400k rpm

All metal components

Ran a few minutes.

Turbine blades melted!

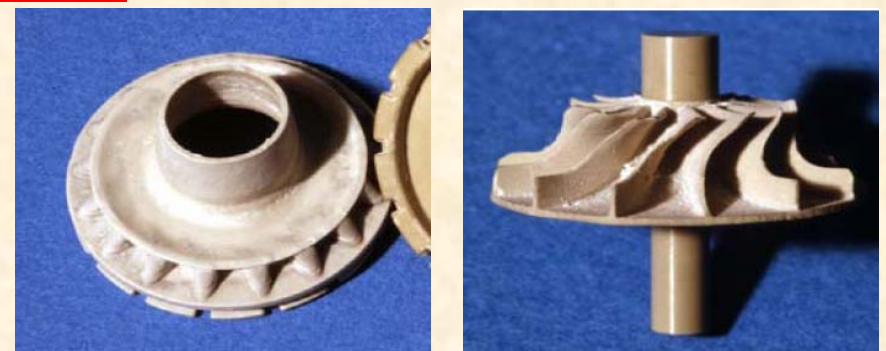


M-DOT micro-turbine engine

1998: DARPA – M-Dot – Stanford – Carnegie Mellon project

Replace the inlet nozzle to improve specific thrust density.

- Inlet nozzle: major ceramic part. Tested in 1,250°C gas
- 7% performance (thrust/weight) improvement expected
- Ceramic turbine built but not tested.



Silicon nitride inlet nozzle and turbine

Figures and text: Kang, S., 2001, Ph.D dissertation, Stanford Univ. & Personal communication with Kang, S.

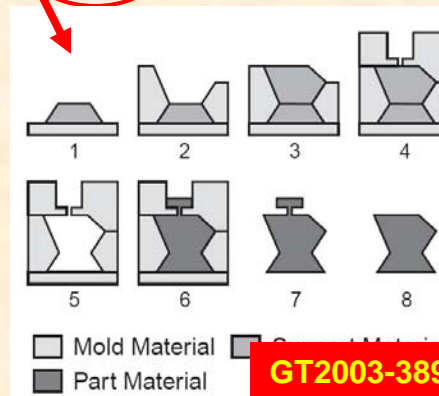
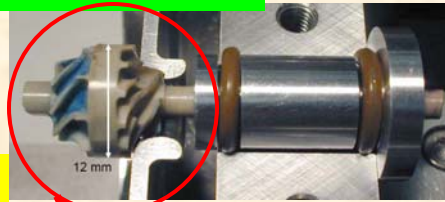


MTM materials & fabrication

Fabrication

- Mold SDM
- Precision 3D Milling
- MEMS

Mold SDM process



GT2003-38933

3D Milling

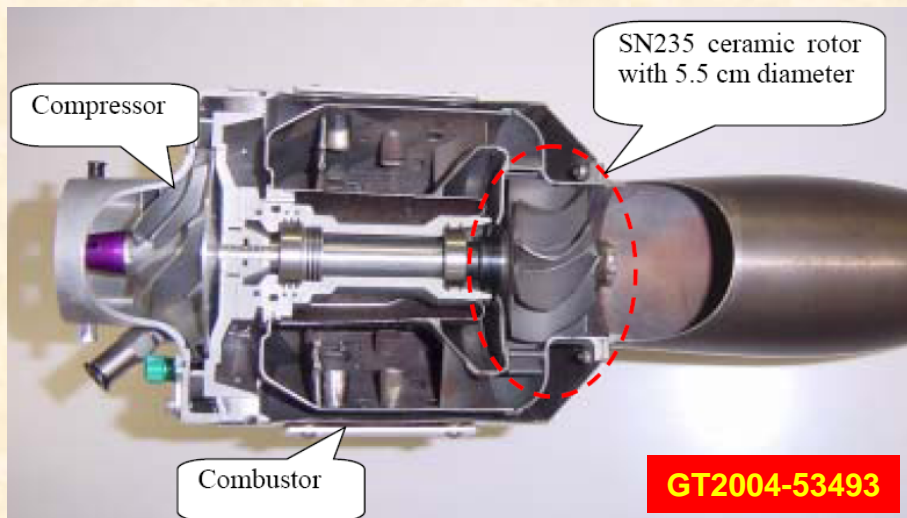


GT2003-38151

DRIE process



GT2003-38866



GT2004-53493

Materials & Reliability

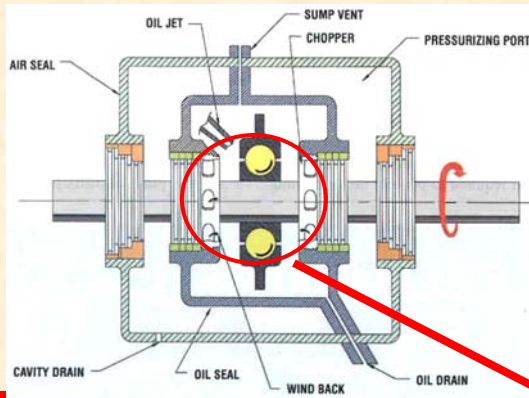
- High temperature durability
- Light weight



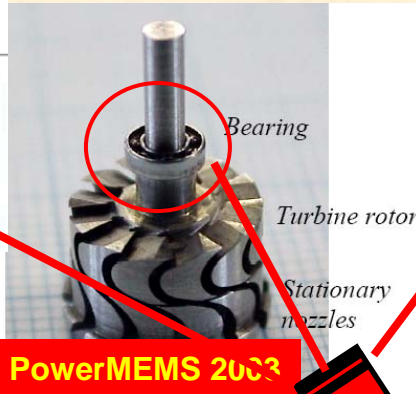
Available Bearing Technologies

Rolling element bearings

- Low temperatures
- Low DN limit ($< 2 \text{ M}$)
- Need lubrication system

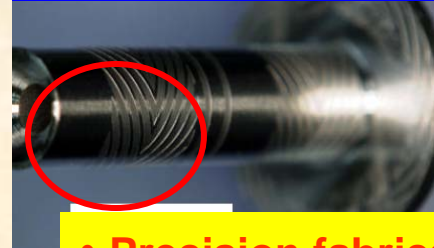


AIAA 2004-4189



PowerMEMS 2003

Herringbone grooved bearing



NICH Center,
Tohoku University

- Precision fabrication process
- Low load capacity and stiffness and little damping

GAS BEARINGS

- Oil-Free
- NO DN limit
- Low friction and power loss
- Thermal management

Gas Foil Bearing



AIAA-2004-5720-984

Flexure Pivot Bearing



GT 2004-53621



MTM – Needs, Hurdles & Issues

**Largest power to weight ratio,
Compact & low # of parts**

**Reliability and efficiency,
Low maintenance**

Extreme temperature and pressure

Environmentally safe (low emissions)

Lower lifecycle cost (\$ kW)



High speed

Rotordynamics & (Oil-free) Bearings & Sealing

Materials

*Coatings: surface conditioning for low friction and wear
Ceramic rotors and components*

Manufacturing

*Automated agile processes
Cost & number*

Processes & Cycles

*Low-NOx combustors for liquid & gas fuels
TH scaling (low Reynolds #)*

Fuels

Best if free (bio-fuels)



Useful websites

NASA Oil-Free Turbomachinery Program

<http://www.grc.nasa.gov/WWW/Oilfree/>

DOE

<http://www.eere.energy.gov/de/microturbines/>

Capstone micro turbine

<http://www.capstoneturbine.com/>

Mohawk Innovative Technology, Inc.

<http://www.miti.cc/>

MIT Gas Turbine Lab.

<http://web.mit.edu/aeroastro/www/labs/GTL/>